PATENT SPECIFICATION

671,567



Date of Application and filing Complete Specification: Feb. 3, 1950. No. 2824/50.

Application made in United States of America on Feb. 4, 1949. Complete Specification Published: May 7, 1952.

Index at acceptance:—Classes 29, G(3x: 15), and 137, C1(b:d).

COMPLETE SPECIFICATION

Improvements in or relating to a Device for Conveying Refrigerated Air through a Duct

We, THE GARRETT CORPORATION, a corporation organised under the laws of the State of California, United States of America, of 9851 Sepulveda Boulevard, Los 5 Angeles, California, United States of America (Assignees of Paul C. Scofield and Frederick H. Grebn), do hereby declare the invention, for which we pray that a patent may be granted to us, and the 10 method by which it is to be performed, to be particularly described in and by the following statement:—

This invention relates generally to air refrigerating systems and particularly to 15 such a system wherein air cooled to a temperature below freezing is conveyed through a duct. Specifically, the invention pertains to a device, adapted for use in connection with an air refrigerating 20 system, for preventing the accretion of entrained ice particles and ice fog within the air flow duct so as to maintain an unrestricted flow of the cooled air through the duct.

25 The present invention provides a device for conveying refrigerated air, comprising an air flow duet having an air expansion region wherein discrete ice particles may be formed in the expanded air; and means for 30 continuously introducing air of relatively higher temperature into said region to impart a latent heat of fusion capable of melting the ice particles.

The present device has particular utility 35 when employed in connection with an air refrigerating system for ventilating and cooling the cabin or cockpit of an aircraft, but it is to be understood that the device may be advantageously utilized in various 40 other air cycle systems for ventilating and refrigerating other areas. In its broadest aspect, this invention is concerned primarily with means for preventing the accretion of ice particles and ice fog within any cold air 45 duct, such as the outlet duct of a cooling turbine. The specific embodiment herein disclosed is, therefore, to be regarded

merely as a typical example of the many possible applications of the device to use,

Under certain operating conditions, it is 50 highly desirable, and more or less essential, that the cabins or cockpits of aircraft be supplied with refrigerated ventilation air. For example, as a general rule refrigera-tion is required only from sea level up to 55 an altitude of approximately 15,000 feet above sea level, such a conditioning of the air being especially desirable when the aircraft is flying in tropical conditions of relatively high temperatures and humidity 60 Various systems have been proposed for carrying out the air refugerating and circulating process. In the case of a circulating process. turbine-powered aircraft, compressed air is bled off from the main compressor for 65 cabin ventilation and/or pressurization purposes. In the case of a reciprocatingengine-powered aircraft, an auxiliary cabin supercharger is driven by the engine and furnishes such cabin air. In either case, the 70 air is relatively hot and compressed to a pressure somewhat above that necessary for the purpose, and in such cases it is common to provide means for cooling or refrigerating the air and reducing its pressure to 75 useful level, the so-called "air cycle" refrigerating systems being the most com-monly used. While such air conditioning systems may comprise various components arranged in innumerable ways, all such 80 systems are based upon the principle of utilizing the expansion of air in a turbine to reduce its temperature substantially below compressor or supercharger outlet values. While other means have been pro- 85 posed for cooling aircraft cabins, the weight and bulk of equipment used in such systems have been prohibitive and for this reason, among others, the air cycle system of cooling is preferred. The present 90 invention is largely concerned with consideration of aircraft in which any source of compressed air is available, the air being forced through a duct for cabin or cockpit

BEST AVAILABLE COPY

BNSDOCID: <GB_____671567A__I_>

[1.

thermostat, all the recirculated air flows through passage 21, since under this condition no ice will form, and the addition of heat upstream of the water separator unit 5 27 would decrease the efficiency of the water separator by lowering the quantity of entrained moisture. In order to prevent the deposition of ice on the duct 12, the duct 12 is provided with fins 44 on its 10 exterior, thus insuring that the wall of the duct will be well warmed by the recirculated air within the passage 20.

Under conditions when notice exists at the turbine discharge outlet, passage 20 will be 15 closed and the water separator will function in the normal manner. The recirculated air then flowing through the annular passage 21 may be mixed with the turbine-cooled air to produce a mixture of unsaturated air 20 which prevents local fogging in the cabin. Regulation of the water separator effectiveness may be accomplished either by use of the valve controlled by-pass opening 45 or by reintroducing water into the air stream 25 by means of the humidifier 30, 31, or by

both means. If the humidity within the cabin or compartment 5 falls below a predetermined value, the humidistat 35 functions to 30 energize the actuator 39 to open the bypass valve 40 which then permits air, which has not been dehumidified, to by-pass the water separator. If this action does not produce sufficient humidity in the com-35 partment 5, the humidistat 35 energizes the actuator 36 which then opens valve 37. In this manner a relatively small quantity of air at relatively high pressure flows from duct 7 through the tube 32 to the aspirator 40 31 where it acts to draw water from the receptacle 29 through the tube 30 and spray it into the duct 13 connected to the compartment, thereby humidifying the air entering the compartment. The valves 25, 37 45 and 40 are capable of modulating in accordance with the conditions. For accordance with the conditions. example, valve 25 may assume a position between fully open for passage 24 and fully open for passage 24', in response to tem-50 perature signals received from the thermostat 33. Likewise, valves 37 and 40 will respectively modulate the flow of air in the tube 32 and the amount of air by-passing the separator 27 in response to signals

55 received from the humidistat 35.

It is within the concept of this invention to so form the ejector-mixers 11 and 42 that they may function as aspirating means for providing the motive power for the air 60 recirculation cycle, thereby eliminating the need for the recirculating far.

It will be apparent that recirculated cabin air is not necessary to the operation of the present device under all flight conditions.

If warm ambient air is available for the 65 purpose, it is possible to substitute such air for the recirculated cabin air to achieve the desired de-icing. This substitution is only possible, of course, when the cabin is unpressurized, as for example, during 70 ground cooling or during flight of the aircraft in the unpressurized range. It is apparent, however, that a source of warm air at any pressure higher than that of the cabin or compartment may be utilized, e.g., 75 ram air, bleed-off air from a turbo-compressor, etc.

In its broadest aspect, our invention resides in providing means for heating the discharged air by mixing therewith air of 80 higher temperature, the result being to raise the temperature of the air to a value above freezing temperature so that ice particles contained in the discharged cooling air are melted. By this means, the ice particles, 85 or ice fog, are prevented from accumulating on the inner surfaces of the discharge duct to create a back pressure which, as will be apparent, would reduce the temperature drop and result in inefficiency of 90 the air refrigerating system or apparatus with which the turbine or supercharger is associated.

It is thus seen that the invention provides means for positively preventing 95 accretion of ice particles within a cold air duct, and that this means includes (a) means for melting the ice which collects on the inside surfaces of the duct and (b) means for mixing warm air with the cold 100 air either at the aspirator or downstream in order that the entrained ice and ice fog will be melted. As will be apparent to one versed in the art, the source of the warm air is not important and it is immaterial 105 whether the cold supersaturated air aspirates the warm air, or vice versa. It is within the concept of this invention to introduce warm air of a higher pressure than the cold air into the de-icer means 110 through the duct 10, in which case the turbine outlet air would be led into the passage 22 to be aspirated by the warm air.

Secondary to the broad concept as discussed above, the invention further 115 provides means, disposed downstream from the air refrigerating means, for separating all or a portion of the water from the cooled air flowing through the duct, so as to vary the humidity of the refrigerated air discharged from the duct in accordance with requirements. As a further feature of the present invention, automatic means are provided for causing some of the air to bypass the water separating means, the by- 125 pass means being controlled by humidistat means. As previously explained, it is a matter of choice as to source of the warm

air employed for de-icing purposes. However, when the refrigerated air is employed as a medium for conditioning the air of an aircraft cabin or other compartment, air in 5 the compartment can be utilized for the purpose, the present invention disclosing means tor recirculating a portion of the compart-

ment air. It will be observed from the foregoing 10 that the present invention provides, in addition to duct de-icing means, an efficient means for maintaining unrestricted flow of the refrigerated air to an aircraft cabin or other compartment at predetermined tem-15 peratures and humidity. The device is particularly adapted for use in aircraft which are operated under climatic conditions of relatively high temperatures and humidities, as are common in the tropics at 20 altitudes up to approximately 15,000 feet. By the use of the present device, the accretion of entrained ice particles and ice fog in the turbine discharge duct of an air refrigerating system is prevented. 25 feature of novelty, the present device functions to remove the entrained ice particles and ice fog by converting the same into water by a melting process, and subsequently removing the water from the air 30 stream by means of a separator.

the cooling energy loss of the air at a minimum. In one form of the apparatus, a selected portion of the ventilation air intro-35 duced into the cabin is recirculated through the apparatus as a means to impart to the ice particles and ice fog the latent heat of fusion necessary to melt them. The present

device also embodies means for maintaining

device also embodies means for reinjecting 40 into the air at a point downstream from the water separator such amounts of water as are necessary to maintain controlled conditions of humidity within the cabin or other

compartment.

What we claim is:-

1. A device for conveying refrigerated air, comprising: an air flow duct having an air expansion region wherein discrete ice particles may be formed in the 50 expanded air; and means for continuously introducing air of relatively higher temperature into said region to impart a latent heat of fusion capable of melting the ice

particles.

2. A device according to claim 1, having an outer fluid conduit surrounding said air flow duct and defining therewith a space through which a heating fluid can flow in contact with said air flow duct so as to 60 warm the walls of said duct to prevent ice particles occurring in the refrigerated air from accumulating on the inner walls of the duct, there being an opening in said air flow duct through which the heating fluid

can flow from said conduit into the duct 65 to mix with the refrigerated air so as to provide the latent heat of fusion necessary to melt discrete ice particles and ice fog in the refrigerated air.

3. A device according to claim 2, where- 70 in said opening comprises an aspirating orifice through which the heating fluid can be drawn into the duct to mix with the

4. A device according to claim 3, where- 75 in the flow of the heating fluid through said orifice creates a low pressure area adjacent said orifice tending to draw the cold air

through said duct.

5. A device according to claim 1, where- 80 in said air flow duct communicates with the interior of a compartment, and having a water separating means disposed in said duct downstream from the point of entrance of said air of higher temperature 85 and adapted to remove the water formed by said melting process so as to dehumidify the air entering the compartment, and humidifying means for introducing water into said air within said duct, downstream 90 from said water separating means.

6. A device according to claim 5, wherein said humidifying means is adapted to inject into the air within said duct, at a point downstream from said water separat- 95 ing means, at least a portion of the water separated from the air by said separating

7. A device according to claim 6, having a receptacle for receiving the water 100 separated from the air by said separator, and wherein said humidifying means includes aspirating means for drawing water from said receptacle and injecting the same 105 into the air stream.

8. A device according to claim 6 or 7, wherein said humidifying or aspirating means is responsive to variations in the humidity of the air within the compartment.

9. A device according to claim 7 or 8, 110 in which said aspirating means includes a tube connected to a source of compressed air and provided with a nozzle disposed in said duct, and a line extends from said receptacle into said nozzle.

10. A device according to claim 5, having by-pass means operative to permit the air discharged from refrigerating means to by-pass said water separating means.

11. A device according to claim 10, 120 wherein said by-pass means is operative in response to variations in the humidity of the air within the compartment.

12. A device according to claim 10, wherein said by-pass means comprises an 125 opening provided between said duct and said separator means and including a valve operative in response to variations in the

115

humidity of the air within the compartment to permit air discharged from the refrigerating means to by-pass said water

separating means.

5 13. A device according to claim 5, having means responsive to variations in the temperature of the air in said duct between refrigerating means and said water separating means for controlling the introduction of said air of higher temperature into said duct.

14. A device according to claims 1 and 5, wherein said air introducing means comprises a warm air conduit communicating
15 with the interior of said cold air duct adjacent the discharge end of refrigerating means, a valve movable within said warm air conduit, and means responsive to variations in the temperature of the air in said 20 cold air duct between the refrigerating

means and said water separating means for controlling the quantity of warm air intro-

duced into said cold air duct.

15. A device according to claim 14, 25 wherein said warm air conduit has a pair of branches through which warm air can flow, the first of said branches communicating with the interior of said cold air duct adjacent the discharge end of the refrigeration of means and the second of said branches communicating with the interior of said cold air duct downstream therefrom, warm

air introduced into said cold air duct through said first branch imparting to ice 35 particles within said cold air duct the latent heat of fusion capable of melting the particles; said water separating means being disposed in said cold air duct between the points of communication of said

40 branches and adapted to remove water formed by said melting process so as to dehumidify the air, and said valve in said warm air duct being operable in response to variations in the temperature of the air

45 in said cold air duct, upstream from said water separating means, for automatically controlling the amount of warm air introduced into said cold air duct through either of said branches.

50 16. A device according to any of claims 5 to 15, having means responsive to variations in the humidity of the air in the compartment for reinjecting water separated from the air stream by said water 55 separating means into the air downstream

from said water separating means.

17. A device according to any of the preceding claims, comprising a cold air duct axially aligned with a discharge duct for 60 refrigerating means and having an end surrounding the discharge duct in spaced relation thereto to provide a first annular air passage therebetween, said cold air duct receiving and conducting cold air dis-

charged by the refrigerating means, an 65 outer air duct surrounding said cold air duct in spaced relation thereto and having an end communicating with the interior of the compartment through which air discharged from the refrigerating means can 70 enter the compartment, one end of said cold air duct cooperating with said discharge duct to define a first annular ejectormixer therebetween, said outer duct cooperating with the other end of said cold 75 air duct to define a second annular ejectormixer therebetween; a partition between said cold air duct and said outer duct and disposed between said first and second ejector-mixers and dividing the space be- 80 tween said cold air duct and said outer duct into two annular passages, the first of said passages leading to said first ejector-mixer and the second of said passages leading to said second ejector- 85 mixer, a warm air conduit having a first branch communicating with said first passage and a second branch communicating with said second passage, warm air in said passages being adapted to flow into 90 the cold air stream at said ejector-mixers, the warm air entering said cold air duct through said first ejector-mixer imparting to any ice particles within said cold air duct the latent heat of fusion capable of 95 melting the particles, a water separating means disposed in said cold air duct between said first and second ejector-mixers and adapted to remove water formed by said melting process so as to dehumidify the air 100 entering the compartment, and a valve in said warm air conduit operable in response to variations in the temperature of the air in said cold air duct, upstream from said water separating means, for controlling the 105 amount of warm air passing through said branches.

18. A device according to claim 17, in which said warm air conduit communicates with the interior of the compartment and 110 is adapted to recirculate air introduced into the compartment for the purpose of melting the ice particles within said cold air duct.

19. A device according to claim 17, in 115 which said outer warm air conduit surrounds said discharge duct and defines therewith an annular space through which the warm air passes to said first ejectormixer and contacts said discharge duct to 120 warm the walls thereof so as to melt ice particles tending to accumulate thereon.

20. A device according to claim 14 or 17, having air propelling means and air filtering means within said warm air 125 conduit.

21. A device according to claim I, wherein the air flow duct comprises a

BNSDOCID: <GB_____671567A__I_>

refrigerated air discharge duct and a cold air duct for receiving the discharged refrigerated air, said cold air duct having an end surrounding said discharge duct and 5 providing an opening through which the air of higher temperature may flow into said cold air duct.

22. A device according to claim 2 or 20, wherein said discharge duct, cold air duct 10 and outer fluid conduit are in concentric relation, and said opening defines an annular passage.

23. A device according to claim 1, characterised by the air expansion region 15 including a turbine for expanding and

cooling the air.

24. A device constructed and adapted to operate substantially as herein described with reference to the accompanying drawing.

20

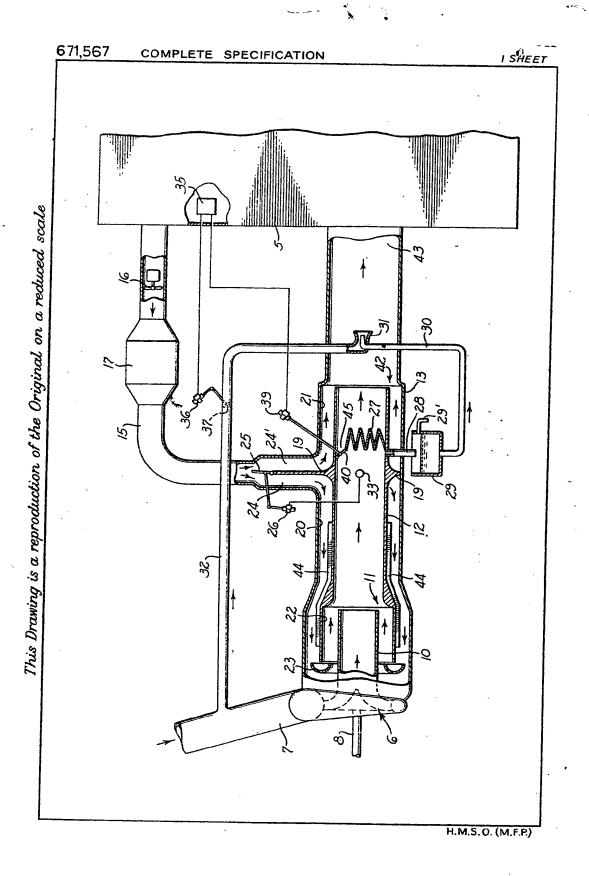
For: THE GARRETT CORPORATION.

STEVENS, LANGNER, PARRY & ROLLINSON,

Chartered Patent Agents, 5/9, Quality Court, Chancery Lane, London, W.C.2,

and at 120, East 41st Street, New York, 17, New York, U.S.A.

Redhill: Printed for Her Majesty's Stationery Office, by Love & Malcomson, Ltd.—1952. Published at The Patent Office, 25, Southampton Buildings, London, W.C.2, from which copies may be obtained.



BNSDOCID: <GB_____671567A_ 1 >

This Page is Inserted by IFW Indexing and Scanning Operations and is not part of the Official Record

BEST AVAILABLE IMAGES

Defective images within this document are accurate representations of the original documents submitted by the applicant.

Defects in the images include but are not limited to the items checked:

BLACK BORDERS
THIS PAGE BLANK (USPIC) Sadis no mottod, 907 ta 770 tu) agami I work to the same of the
☐ FADED TEXT OR DRAWING
☐ BLURRED OR ILLEGIBLE TEXT OR DRAWING
☐ SKEWED/SLANTED IMAGES
☐ COLOR OR BLACK AND WHITE PHOTOGRAPHS
☐ GRAY SCALE DOCUMENTS
☐ LINES OR MARKS ON ORIGINAL DOCUMENT
☐ REFERENCE(S) OR EXHIBIT(S) SUBMITTED ARE POOR QUALITY
□ OTHER:

IMAGES ARE BEST AVAILABLE COPY.

As rescanning these documents will not correct the image problems checked, please do not report these problems to the IFW Image Problem Mailbox.

THIS PAGE BLANK (USPIU,